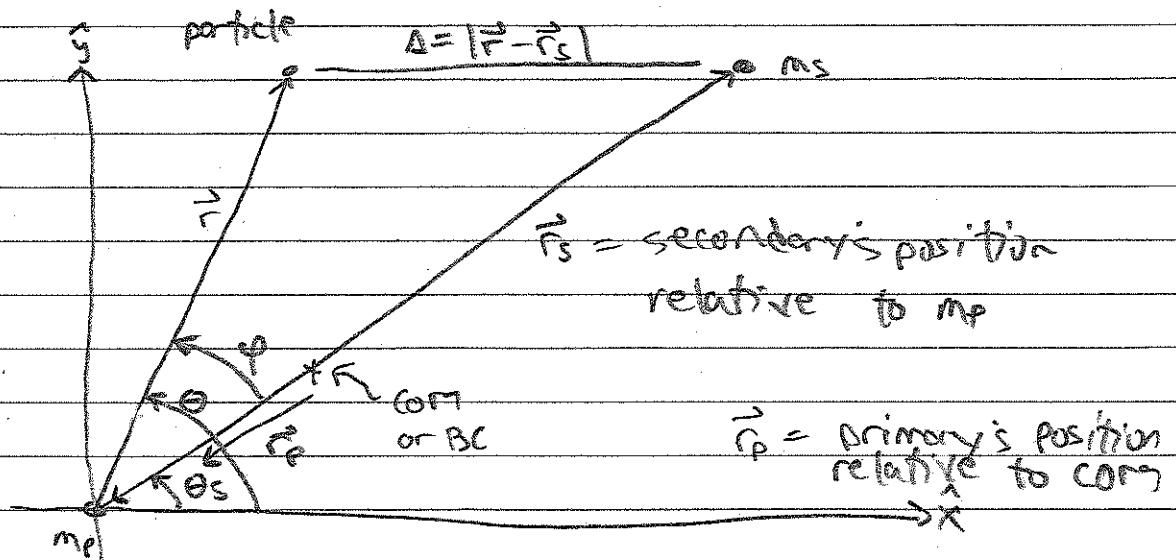


5 Nov 2013

Chapter 6: Lindblad Resonance

Let's calculate the motion of a small particle orbiting near a secondary planet's Lindblad resonance



Recall the Fourier expansion of the secondary's orbit (pg 18 in previous lecture notes):

$$\ddot{\vec{r}}_S(r, \varphi) = -\frac{Gm_2}{\Delta} = \frac{1}{2}\dot{\vec{r}}_p(r) + \sum_{m=1}^{\infty} \vec{\phi}_m(r) \cos(m\varphi)$$



m^{th} Fourier coefficient

Assume system is coplanar, $z=0$

Solve for coefficients ϕ_n : How?

Multiply above by $\cos m' \varphi$ and integrate over φ :

$$\int_{-\pi}^{\pi} E_2(r, \varphi) \cos m' \varphi d\varphi$$

$m' = \text{some integer}$
to be specified
later

$$= \int_{-\pi}^{\pi} d\varphi \cos m' \varphi \left[Y_2 \Phi_0(r) + \sum_{n=1}^{\infty} \Phi_n(r) \cos(m n \varphi) \right]$$

If $m=0$ the LHS = $\pi \Phi_0(r)$

$$\text{for } m \geq 1 \quad \text{LHS} = \sum_{n=1}^{\infty} \Phi_n(r) \int_{-\pi}^{\pi} \cos(m' \varphi) \cos(m n \varphi) d\varphi$$

$$\frac{1}{2} \cos[(m+n)m]\varphi]$$

$$+ \frac{1}{2} \cos[(m'-m)m]\varphi]$$

What is the
integral of
this term?
this?

$$\text{LHS} = \sum_{n=1}^{\infty} \Phi_n(r) \int_{-\pi}^{\pi} \frac{1}{2} \delta_{mn}$$

$$= \pi \Phi_m(r)$$

$$\text{so } \phi_{m1}(r) = \frac{1}{\pi} \int_{-\pi}^{\pi} \mathbb{E}_2(r, \varphi) \cos(m\varphi) d\varphi$$

$$= \frac{2}{\pi} \int_0^\pi \mathbb{E}_2(r, \varphi) \cos(m\varphi) d\varphi$$

since $\mathbb{E}_2(-\varphi) = \mathbb{E}_2(\varphi)$

β even

The secondary's potential is

$$\mathbb{E}_2 = -\frac{Gm_s}{\Delta} = -\frac{Gm_s}{\sqrt{r^2 + r_s^2 - 2rr_s \cos\varphi}}$$

$$\text{so } \phi_{m1}(r) = -\frac{Gm_s}{r_s} \frac{2}{\pi} \int_0^\pi \frac{\cos(m\varphi) d\varphi}{(1 + \beta^2 - 2\beta \cos\varphi)^{1/2}}$$

$$\beta = \frac{r}{r_s}$$

what is this?

$$\text{so } \phi_{m1}(r) = -\frac{Gm_s}{r_s} b_{12}^{(m)}(\beta)$$



Laplace
coefficient

Now let's account for the indirect potential Σ_i :

Newton's 2nd Law: $\ddot{\vec{r}}_{\text{com}} = -\nabla (\Sigma_p + \Sigma_i)$

in an inertial ref' frame, where

\vec{r}_{com} = particle's position relative to COM (ie E.C.)

$$= \vec{r} + \vec{r}_p$$

so $\ddot{\vec{r}} = \ddot{\vec{r}}_{\text{com}} - \ddot{\vec{r}}_p$ = where $\ddot{\vec{r}}_p = \frac{Gm_s}{r_s^3} \hat{r}$

= particle's
acceleration

due to m_s

Assignment #5

problem 6.1:

Show that $\ddot{\vec{r}}_p = -\nabla \Sigma_i = \frac{Gm_s}{r_s^3} \hat{r}$

where $\Sigma_i = \frac{Gm_s}{r_s^3} \hat{r} \cdot \hat{r}_s =$ indirect potential

$$= \frac{Gm_s}{r_s^3} r r_s \cos \phi$$

$$= \frac{Gm_s}{r_s} r \cos \phi$$

so the particle's EOM in the non-inertial ms-centered reference frame is

$$\ddot{\vec{r}} = -\nabla(E_p + \vec{E}_s + \vec{E}_i)$$

where \vec{E}_i is an effective potential

that accounts for the additional fictitious forces that appear when you use a non-inertial coordinate system.

So the total potential on the particle due to all of the secondary's perturbations is

$$\vec{E}_s + \vec{E}_i = \frac{1}{2} \Phi_0 + \sum_{n=1}^{\infty} \varphi_m(r) \cos(m\theta) + \frac{Gm_s}{r} \beta \cos\theta$$

$$\text{where } \varphi_m(r) = -\frac{Gm_s}{r_s} b^{(m)} \gamma_2(\beta)$$

The m's in the Fourier expansions are the perturbation's azimuthal wavenumber

i.e. the perturbing forces cycle m-times as you travel around 2π radians in longitude.

What is the azimuthal wavenumber for \vec{E}_i ?

$$m=1$$

So we can fold \vec{E}_i into the $m=1$ term in the Fourier expansion:

$$\vec{E}_s + \vec{E}_i \rightarrow \vec{E}_i = \frac{1}{2} \vec{e}_B + \sum_{m=1}^{\infty} \phi_m(r) \cos(m\varphi)$$

where $\phi_m = -\frac{Gm\mu_0}{B} \left[b \gamma_2(B) - \delta_m(B) \right]$

$$f \quad f$$

These two terms
are usually comparable
in magnitude

If we ignored \vec{E}_i , our results
would have been in error by factor ~ 2 ,
but only for a particle orbiting near
the secondary's m=1 Lindblad resonance

(also known as the 2:1 or 1:2 mean motion resonance)

Motion of a particle near a Lindblad resonance (L_E)

The particle is perturbed by an infinite number of terms in the secondary's Fourier -expand potential $\mathbb{E}_S = \mathbb{E}_0 + \mathbb{E}_1 + \mathbb{E}_2 + \mathbb{E}_3 + \dots + \mathbb{E}_{\text{int}}$

but the following will show that if the particle is near an n^{th} LR, the particle behaves as if it were only perturbed by the n^{th} term \mathbb{E}_n

so write

$$\mathbb{E}_S(r, \theta) = \mathbb{E}_n = c_n(r) \cos(n\theta)$$

and ignore the other terms

it's not that the other terms are small and negligible (they are not)

it's that the \mathbb{E}_n term is resonant
ie the forcing frequency associated with \mathbb{E}_n is nearly equal to the particle's natural oscillation frequency

resonant

The particle's response to all those n terms in \mathbb{E}_S is

$$\leftarrow \text{due to } \mathbb{E}_2 \quad \leftarrow \text{due to } \mathbb{E}_n$$

$$r = r_0 + r_1 + r_2 + \dots + r_n + \dots$$

↑
only this one is large

$$\text{So } r(t) = r_0 + r_m(t)$$

which is why we can write

$$\begin{aligned} \vec{r}(r, \theta) &\approx \phi_m(r) \cos(m\theta) \\ &= \text{Re}[\phi_m(r) e^{im(\theta-\Theta)}] \end{aligned}$$

when using complex notation,

and $\varphi = \theta - \Theta = \text{particle's long/lat}$
relative to us

$$\text{and } e^{im\theta} = \cos m\theta + i \sin m\theta$$

The particle's Equation of Motion (EOM):

in polar coordinates, see Ch 5 notes pg 2

$$\hat{r}: \ddot{r} - r\dot{\theta}^2 = -\frac{d}{dr} [\Phi_p + \phi_m e^{im(\theta-\Theta)}]$$

$$\hat{\theta}: \text{and } \frac{1}{r} \frac{d}{dt} (r^2 \dot{\theta}) = -\frac{1}{r} \frac{d}{dr} (\Phi_p + \Phi_s) = -\frac{i m}{r} \phi_m e^{im(\theta-\Theta)}$$

where I've dropped $\text{Re}()$ = keep real part of ()

\Rightarrow only the real part of these equations are
to be preserved, their imaginary parts
are to be ignored

so the above eqn is

$$\frac{1}{r} \frac{d}{dr} (r^2 \dot{\theta}) = \frac{m}{r} q_m \sin[m(\theta - \phi)]$$

the solution:

as before, write

$$r(t) = r_0 + r_i(t)$$

$$\theta(t) = \theta_0 + \omega_{\text{rot}} t + \theta_i(t)$$

and assume

The noncircular
deviations

$$r_i, \theta_i \text{ are small: } |r_i| \ll r_0 \quad |\theta_i| \ll 1$$

which allows us to linearize the EOM

Let's assume secondary's orbit is circular,

$$\text{so } r_s(t) = a_s \text{ and } \theta_s(t) = \omega_s t$$

↖ this assumes

m_s is on \hat{x} axis
when $t=0$

The acceleration from m_s is also small since $m_s \ll m_p$,

$$\text{so } \left| \frac{d\theta_m}{dt} \right| \sim O\left(\frac{\rho_m}{r}\right) \ll \left| \frac{d\theta_p}{dt} \right| = \frac{Gm_s}{r^2} = \Omega^2 r$$

$$\text{so } |\dot{\theta}_m| \ll (\Omega r)^2$$

Now recall $h = r^2\dot{\theta}$ = specific angular momentum

so the $\hat{\theta}$ EOM is

$$\frac{dh}{dt} = -im\theta_m e^{im(\theta-\theta_0)}$$

this is
a small
quantity

so we only need
this to lowest order

$$e^{im\theta} = e^{im(\theta-\theta_0)} = e^{im(\theta_0 + \omega_0 t + \theta_1 - \omega_1 t)}$$

$$\approx e^{[m\theta_0 + m(\omega_1 - \omega_0)t]}$$

set $w_m = m(\omega_1 - \omega_0)$ = particle's
Doppler-shifted
forcing frequency

$$\text{so } im(\theta - \theta_0) = m\theta_0 + w_m t \text{ to lowest order}$$

$$\text{and } \frac{dh}{dt} \approx -im\theta_m e^{im(m\theta_0 + w_m t)}$$

which is integrable:

$$h(t) = h_0 - \frac{m\omega_m}{\omega_m} e^{i(\omega_0 + \omega_m t)}$$

↓
integration constant

Note $h = r^2\dot{\theta} = (r_0 + \gamma)^2 (\dot{\theta}_0 + \dot{\theta}_1)$
 $= r_0^2 \dot{\theta}_0 + 2r_0 r_0 \gamma + r_0^2 \dot{\theta}_1$

=! the above

so choose $h_0 = r_0^2 \dot{\theta}_0 =$ specific angular momentum of circular orbit of radius r_0

$$\text{so } \dot{\theta}_1 = -\frac{2r_0}{r_0} \gamma - \frac{m}{r_0^2 \omega_m} \text{ cm e}^{i(\omega_0 + \omega_m t)}$$

shorthand
for $i(\omega_0 + \omega_m t)$

This will be inserted
into the \vec{r} form:

\hat{r} form: $r = r_0 + r_1$ where $|r_1| \ll r_0$

$$\ddot{r}_1 - (\omega_0 + \dot{\theta}_1)(\omega_0 + \dot{\theta}_1)^2 = -\frac{d}{dr} \left(\frac{\partial \Phi_p}{\partial r} + q_m e^{im\phi} \right)$$

we need DR RHS to 1st order in

the small quantities

evaluate derivatives at $r=0$

$$\frac{d\Phi_p}{dr} = \left. \frac{d\Phi_p}{dr} \right|_{r_0} + r_1 \left. \frac{d^2\Phi_p}{dr^2} \right|_{r_0} + O(r_1^2)$$

while $\frac{d\Phi_m}{dr} e^{im\phi}$ is 1st order small

$$\text{so } \ddot{r}_1 - (\omega_0 + \dot{\theta}_1)(\omega_0^2 + 2\omega_0 \dot{\theta}_1) \approx -\frac{d\Phi_p}{dr} \Big|_{r_0} - r_1 \left. \frac{d^2\Phi_p}{dr^2} \right|_{r_0}$$

$$\ddot{r}_1 - (\omega_0^2 + 2\omega_0 \dot{\theta}_1) r_1 - \frac{d\Phi_m}{dr} e^{im\phi}$$

$$\ddot{r}_1 - \omega_0^2 r_1 - \omega_0^2 r_1 - 2\omega_0 \dot{\theta}_1 r_1 + \left. \frac{d\Phi_p}{dr} \right|_{r_0} + r_1 \left. \frac{d^2\Phi_p}{dr^2} \right|_{r_0}$$

$$= -\frac{d\Phi_m}{dr} e^{im\phi}$$

$$\ddot{r}_1 - \left(\omega_0^2 - \frac{d\Phi_p}{dr} \right) r_0 + \left(-\omega_0^2 + \frac{d^2\Phi_p}{dr^2} + 4\dot{\theta}_1^2 \right) r_0 r_1$$

$$= - \left(\frac{d\Phi_m}{dr} + \frac{2m\omega_0 \dot{\theta}_1}{r_0 \omega_m} \right) r_0 e^{im\phi}$$

The RHS involves all perturbations due to m_2 ,
They are all oscillatory.

what does that tell us about the first
parentheses?

$$\Rightarrow r^2 = \frac{1}{r} \frac{d\dot{\theta}}{dr}$$

This is the familiar formula for
angular velocity of a circular orbit

The next () is

$$3r^2 + \frac{d\dot{\phi}}{dr^2} = ? = \kappa^2 = 4R^2 + r \frac{d\dot{r}^2}{dr}$$

↑ epicyclic frequency²

lets set $\tilde{F}_{\text{in}}(r) = -\frac{dm}{dr} - \frac{2m\omega}{r\omega_m} \dot{\phi}_m$

= secondary's forcing function

so the EOM is

$$\ddot{r} + \kappa^2 r = -\tilde{F}_{\text{in}}(r) e^{i(\text{const}+wt)}$$

is real cos (const+wt)

since
 $\dot{\phi}_m$ is

what kind of DEQ is this?

forced simple harmonic oscillator

The solution has two parts:

$$r(t) = \text{free}(t) + \text{forced}(t)$$

y_p

y_f

part of solution

The solution to

entirely

EOM when

due to me

in your

RHS = 0

DEQ

homogeneous

particular

class, these

solution

solution

are the

phase

$$\text{what is } \text{free}(t) ? = -R \cos(K_0 t + \phi_0)$$

$$= -R e^{i(K_0 t + \phi_0)}$$

and

$\text{forced}(t)$ will resemble

$$\text{forced}(t) = -R e^{i(m\theta_0 + \omega_0 t)}$$

↑ insert into EOM

to solve for the

amplitude of

particle's forced motion

$$-(\omega_m)^2 R e^{i\alpha t} - k^2 R e^{i\alpha t} = -\Phi_m e^{i\alpha t}$$

$$\text{so } -(k^2 - \omega_m^2) R = -\Phi_m$$

set $D(r) = k^2(r) - \omega_m^2(r)$

$$\text{so } R(r) = \frac{-\Phi_m(r)}{D(r)}$$

= amplitude of the particle's
radial motion excited by Φ_m

The particle's forced eccentricity is

$$e = \left| \frac{R}{a} \right| = \left| \frac{\Phi_m}{r D} \right|$$

resonance location

where is the resonance?

where $D(r) = 0$

particle's distance from

resonance in frequency² units

let r_r = resonance radius

$$D(r_r) = K^2(r_r) - \omega_m^2(r_r) = 0$$

$$\Rightarrow K(r_r) = \pm \omega_m(r_r)$$

$$\begin{matrix} \uparrow \\ e = +1 \text{ or } -1 \end{matrix}$$

recall $\omega_m(r_r) = m [\omega_2(r_r) - \omega_S]$ = Doppler shift
Lan^{gular} force freq

$$\text{so } K(r_r) = \pm m [\omega_2(r_r) - \omega_S]$$

is the equation for radius of resonance

Suppose the system is Keplerian
(ie planet orbits a star)

$$\text{then } K = \omega_2 \text{ and } \omega_2 \propto r^{-3/2}$$

$$\text{and } (\pm m + 1) \omega_2 = \pm m \omega_S$$

$$\text{or } \frac{\omega_2}{\omega_S} = \frac{\pm m}{\pm m - 1} = \frac{m}{m - 1}$$

$$\left(\frac{a_S}{r} \right)^{3/2} = \frac{m}{m - 1}$$

$$\text{or } \frac{r}{a_S} = \left(\frac{m - 1}{m} \right)^{2/3} = \left(1 - \frac{1}{m} \right)^{2/3}$$

$$r_r = \left(1 - \frac{1}{m} \right)^{2/3} a_S$$

(R)

= radius of the m^{th} Lindblad resonance

resonance with $\epsilon=+1$ are inner LR (ILR)
since $r_r < a_s$

and $\epsilon=-1$ is an outer LR (OLR)
since $r_r > a_s$

see Fig 6.2

Note that the $m \geq 1$ resonances tend to pile up at $r_r \approx a_s$ at secondary's orbit, which is also known as the corotation circle (cc) since a particle there would corotate with the secondary

Note that when m is large,
the resonances are very close to each other.

In that case, our assumption that a particle responds only to a single resonance breaks down.

These LR resonances are also known as mean motion resonances that satisfy:

$$\frac{P}{P_S} \rightarrow \frac{n}{n_S} = \frac{m}{m-t}$$

or commensurability resonances since

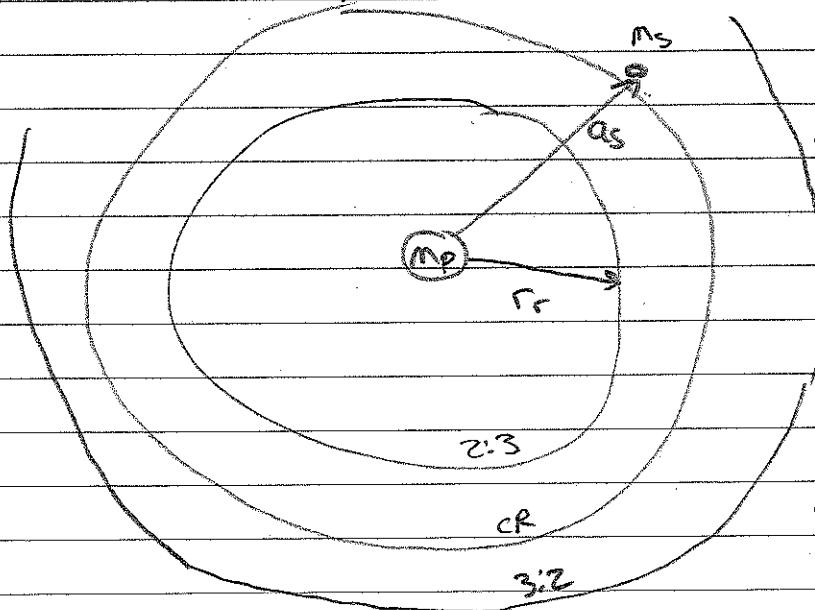
$$\frac{P}{P_S} \rightarrow \frac{P_S}{P} = \frac{m}{m-t}$$

The orbit periods are ratios of integers

ex. The $m=2, \epsilon=+1$ IRL

has $\frac{P}{P_S} = \frac{m-t}{m} = \frac{2}{3}$

so the particle's orbit period is $\frac{2}{3} \times$ secondary's



where

$$r_r = \left(+\frac{\epsilon}{m} \right)^{1/3} a_S$$

$$= 0.76 a_S$$

is sometimes called the
2:3 mean motion resonance

while the $m=2, \epsilon=-1$ OLR is at $r_r = 1.21 a_S$

which has $\frac{P}{P_S} = \frac{3}{2}$ ie the 3:2

The particle's forced eccentricity due to the secondary's m th LR:

$$R(r) = -\frac{\Omega_m(r)}{D(r)}$$

$$\text{where } D(r) = r^2 - w_m^2$$

if the particle is near exact resonance where $\delta = \alpha_r$, then write

$$D(\delta) \approx (r_0 - r_f) \left. \frac{dD}{dr} \right|_{r=r_0} + \text{small} \quad (\text{typo in text})$$

Then set $x = \frac{r_0 - r_f}{r_f}$ = particle's fractional distance from exact resonance

$$\text{so } D(\delta) \approx x \mathcal{D} \text{ where}$$

$$\text{where } \mathcal{D} = \left. r \frac{dD}{dr} \right|_{r=r_f}$$

In Assignment #5 problem 6.4

$$\text{you will show that } \mathcal{D} = 3E(m-\epsilon) \Delta_0^2$$

Assign #5

problems 5.6, 5.7, 6.1, 6.4, 6.5, 6.10

due Tues Nov 19

and in problem 6.5 you show that

$$-\mathcal{E}_m(\phi) = -\frac{\partial \mathcal{S}_m}{\partial r} - \frac{2m\phi_0}{r_{\text{sw}}} \phi_m$$

$$\text{where } \phi_m = -\mu_s \left(b^m \gamma_2(\beta) - \xi_m(\beta) \right) (2\pi)^2$$

$\downarrow \quad \mu_s = \frac{m_e}{m_p}$

$$\Rightarrow \Im m = \epsilon f_m^e \mu s \cos^2$$

$$[f_m = Eqn \ 6.27]$$

= combination of Laplace coefficients
evaluated at resonance,
Tabulated in 6.1

for $1 \leq m \leq 10$

Lfns & W

so the particle's forced eccentricity

$$e(n) = \left| \frac{\frac{2\pi n}{D}}{0} \right| = \frac{f_m^{\text{MS NR2}}}{\times 3(m-t) f_{\text{R2}}^2}$$

$$\text{so } e(x) = \frac{Mf_m^e}{3x(m-t)} = \text{ eccentricity } \text{ VCSNS}$$

$3x(m-t)$ VERSUS

602

fractional distance x
from resonance

example: consider an asteroid orbiting
0.1 AU away from Jupiter's
 $m=3$ $\epsilon=+1$ ILR

what is e excited by this resonance?

Jupiter has $\mu_J = 10^{-3}$ and $a_J = 5$ AU

$$\text{so } x_{\text{res}} = \left(1 - \frac{\epsilon}{m}\right)^{2/3} a_J = 3.8 \text{ AU}$$

= resonance radius

$$\text{Take } \text{so } x = \frac{0.1 \text{ AU}}{r_c} = 0.026$$

Table 6.1: $f_m^{\epsilon}=3$

$$\text{so } e(x) = \frac{\mu_J f_m^{\epsilon}}{3(m-t)x} \approx 0.02$$

what if asteroid was 0.01 AU away
from resonance?

$$e(x) = 0.2$$

\Rightarrow very large e 's can get excited
when close to a LR

Forced eccentricity:

recall $r_i(t) = r_{\text{free}}(t) + r_{\text{forced}}$

= particle's displacement from circular orbit

where $r_{\text{free}}(t) = -R \cos(\Omega t + \phi)$

ps semimajor axis

$$= -R_{\text{free}} \cos(\Omega t + \phi)$$

↑ corresponds to
eccentricity elliptic motion
associated in the guiding
w/ the particle's center approx
free motion = $\frac{R}{r_0}$ when $r_{\text{free}} \ll 1$

see 13.10 notes
↓

while $r_{\text{forced}}(t) = -R \cos(m\theta_0 + \omega_m t) \approx -R \cos m(\theta - \theta_0)$

$$= -r_{\text{forced}} \cos(m\theta_0 + \omega_m t)$$

we call this

the ps forced

eccentricity

even tho the ps forced

motion is NOT

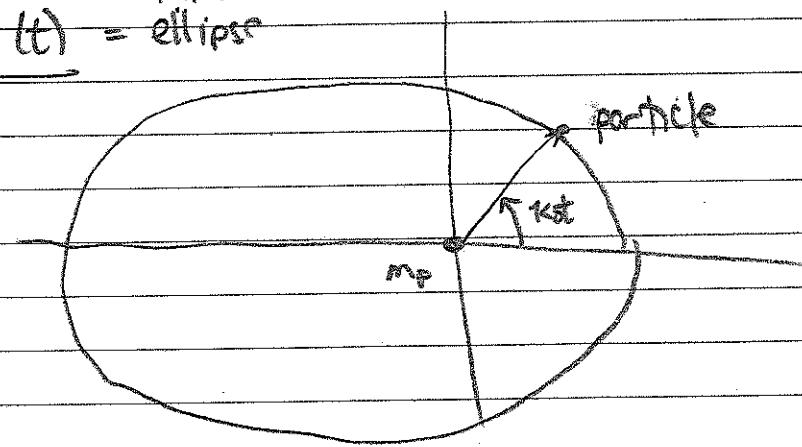
a keplerian ellipse

actually, $r_{\text{forced}} = \frac{R}{r_0} = \text{ps fractional radial displacement}$
due to m_s

Compare the particle's free & forced motions:

assume $K_0 = \omega_0 = \text{mean motion}$ and set $\psi = 0$

$r_{\text{free}}(t) = \text{Keplerian ellipse}$



$r_{\text{forced}}(t)$ assuming particle orbits at $m=2$

$$r_{\text{forced}} = -R \cos m(t + \psi)$$

choose coordinate system
to rotate w/ m_2

$$\text{so } \theta_2 = 0$$

assuming

$$\text{if } f=+1 \text{ ICR } m_2=0$$

$$\text{if } f=-1 \text{ OLR}$$

$$\vec{x} \text{ so } m_2$$

then longitude $m_2 = \omega_{\text{int}}$

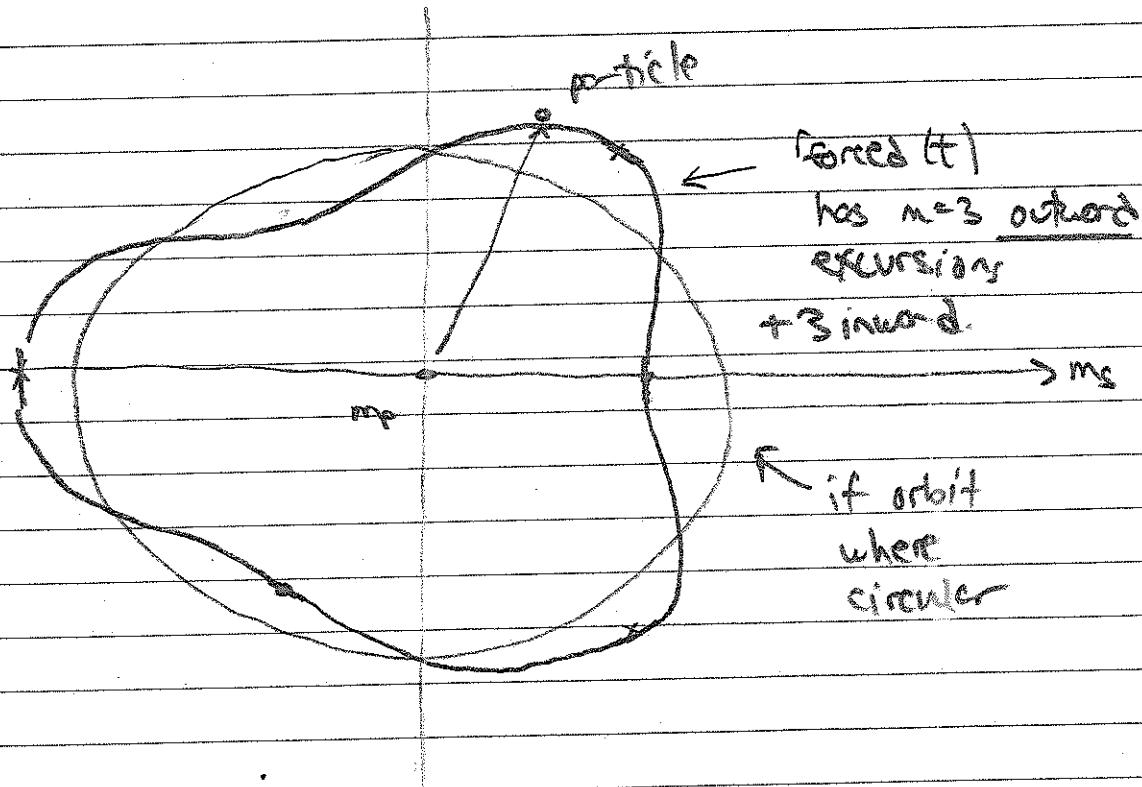
but $\omega_m = \epsilon Kt$ at resonance

$$\text{so } \theta = \frac{\epsilon Kt}{m}$$

where $\epsilon = +1(-1)$

at inner (outer) LR \Rightarrow In the corotating ref' frame, the p's
Note that the longitude increases (decreases) if at inner (outer) LR
p's $m=2$ forced motion is a planet-centered ellipse.

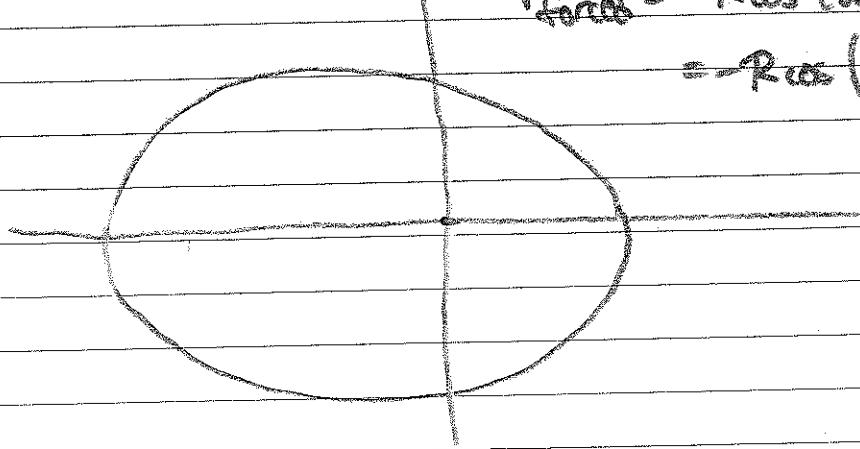
if at $m=3$ LR:



\Rightarrow only the $n=1$ OLR results in an elliptic trajectory that is elliptic

$$\vec{r}_{\text{forced}} = -R \cos(\omega_m t)$$

$$= R \cos(\pi - \omega_m t)$$



This is why the B ring's outer edge

(at Mimas' $m=2$ ILR)

has an $m=2$ saturn-centered shape (mostly)

and why the A ring's outer edge

has $m=7$ shape, due to $m=7$ ILR

with Janus & Epimetheus.

Other resonances:

recall $r_{\text{eff}}(t) \propto R$

m_s 's
= forcing for

$$\text{where } R \propto \Sigma_m = -\frac{\partial \Phi_m}{\partial r} - \frac{2\pi SR}{r m \Omega_m}$$

$\omega_m = m(SR - S_{\text{S}}$) \Rightarrow Doppler shifted forcing freq

so R gets large when $SR = S_{\text{S}}$

This is a corotation resonance (CR resonance)

i.e. on CR circle

This particular CR lies on the secondary's orbit,
at r_{SOS}

and is not particularly important (why?)

however there are other weaker Lindblad & CR resonances that do not lie on the CR circ

and thus occur in nature (ie ring-satellite systems, asteroid belt + Jupiter/Mars etc)

They are associated with the secondary's eccentric motion, which results in additional radial forcing on the particle that is periodic and thus can result in resonant excitation.

The strength of these higher-order LR and CR resonances are weaker than the zeroth-order LR studied here, by factor of $\epsilon_3^{(k)}$ where $|k| = 1$ (or more)

This is assessed in Section 6.2, which shows where these higher order LR and CR resonances are.

see Fig 6.3.

See section 6.22 for derivation of vertical resonances... if particle & m_s are not coplanar, the particle experiences periodic vertical forces from m_s , which can excite the particle's fixed inclination

Resonance Trapping

many planetary bodies are in Lindblad resonances:

Pluto is in a 3:2 resonance w/ Neptune
(ie at Neptune's $n=2$ OLR)

3 of Jupiter's 4 Galilean sat's
(Io, Europa, Ganymede + Callisto)
are in Laplace resonance, ie 1:2:1
resonances

The 4 that
Galileo could
see w/ his
telescope

Saturn's A,B rings w/ Mimas, Janus & Epimetheus

Mimas is in 2:3 resonance w/ Enceladus

Uranian sat's are resonant:

Umbriel - Titania - Oberon
1:2 2:3

HST discovered 3 new sat's at Pluto,
all seem to be resonant w/ Charon

Exoplanet systems

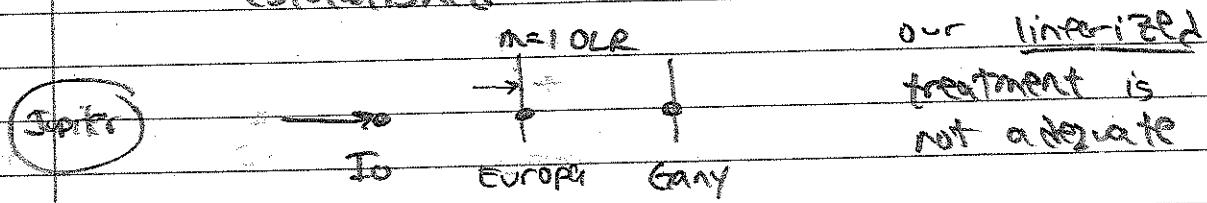
GJ 876 has 2 Jupiter mass planets in 2:1
+ many others

Probably due to N's outward migration
+ resonance capture

to solve for this evolution,
requires a nonlinear theory for particles
motion at resonance

(libration, level curves, conservation of action)

likely explains how Laplace resonance was
established



Alternatively, weak drag force can also
cause a small body's orbit to decay inwards
into a planet's resonance

if planet's perturbation can deposit
enough E and L with the particle
to offset losses due to drag,
the particle gets trapped at the resonance:

ex: PR drag can cause dust to spiral
in and get trapped at LR's

(astronomers are looking for this in
circumstellar debris disks, but no
unambiguous detections yet)

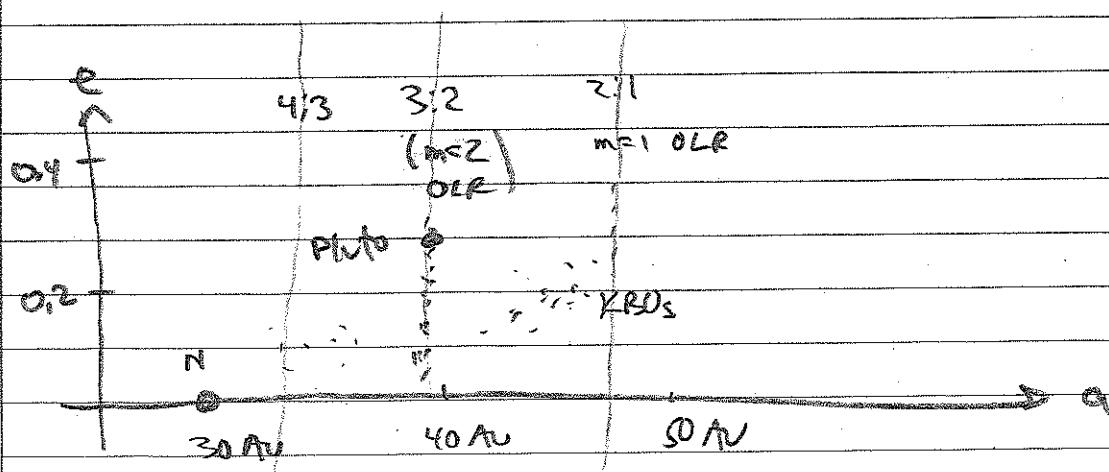
How do bodies get into resonance?
do they form in resonance?

Pluto's $e=0.25$. Did it form in this orbit?

2 easy ways to establish resonance:

resonance sweeping: the secondary's orbit migrates radially, its LR resonances also migrate, or "sweep". This can cause small bodies to get trapped at Lindblad resonances

Neptune, Pluto, & The Kuiper Belt:



How did this system come to be?

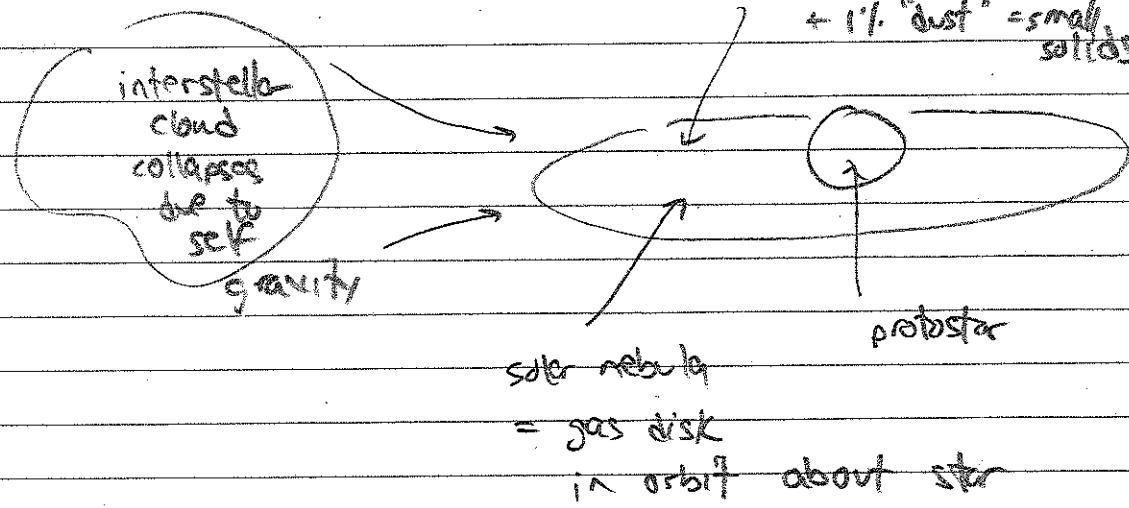
also: gas drag on planetesimals due to solar nebula gas drag can deliver planetesimals to resonance.

This kind of resonant trapping can be solved using linear theory.

The solar nebula & origin of planets,
in 2 minutes

probably 99% gas (H_2)

+ 1% "dust" = small solids



dust assemble into planetesimals \rightarrow protoplanets

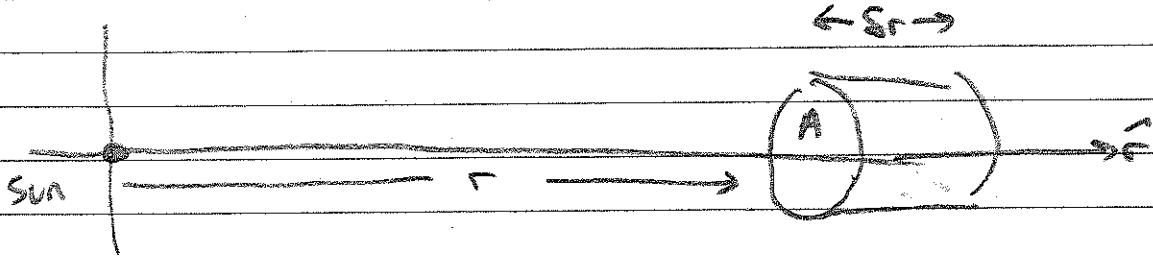
~10% protoplanets accrete nebula gas \rightarrow giant planets

smaller protoplanets \rightarrow terrestrial planets

remaining planetesimals get accreted or scattered by planets, or they survive and persist in asteroid, Kuiper Belts

The solar nebula's sub-Keplerian rotation about the Sun. From Section 3.2.2

consider a parcel of nebula gas of mass δm in cylinder of size $A \times \delta r$:



$\delta m \rho A \delta r = \text{mass of gas parcel}$

nebula density

$$\text{the F term is } \delta m (\ddot{r} - r\dot{\theta}^2) = -\frac{Gm_1 \delta m}{r^2} + A \delta p$$

↑ sun's gravity on δm

where $\delta p = \text{pressure difference}$
across cylinder

$$= p(r) - p(r + \delta r) \approx -\frac{dp}{dr} \delta r$$

$$\text{so } \ddot{r} - r\dot{\theta}^2 = -\frac{Gm_1}{r^2} - \frac{1}{\rho} \frac{dp}{dr}$$

The nebula gas will have settled into circular orbits so $\dot{r} = 0$ and $\dot{\theta} = \Omega$

$$\text{so } \Omega^2 = \frac{GM}{r^3} + \frac{1}{\rho r} \frac{dp}{dr}$$

\sim

negative

since $p(r)$ likely decreases with r

$$\text{set } \Omega_0^2 = \frac{GM}{r^3} = \text{Keplerian angular velocity}^2$$

$$\text{so } \Omega^2(r) = (1 - 2\eta) \Omega_0^2$$

$$\text{where } -2\eta = \frac{d/dr}{\rho r \Omega_0^2}$$

$$\text{or } \eta = \frac{dp/dr}{2\rho r \Omega_0^2}$$

$$\text{so } \Omega(r) = (1 - \eta) \Omega_0 = \text{gas' angular velocity}$$

$$\text{so } \vec{v}_{\text{gas}} = r\Omega\hat{\theta} = (1 - \eta) r\Omega_0 \hat{\theta}$$

↑ when $\eta > 0$,

the gas motion is subkeplerian

so planetesimals orbiting in solar nebula experience "headwind" which exerts aerodynamic drag, causing planetesimal

to spiral inward and into a protoplanet's LR

Calculate τ for planetesimal orbiting at $r = 1 \text{ AU}$:

$$\text{ideal gas law: } p = \frac{\rho k_B T}{m_g}$$

ρ = gas volume density

k_B = Boltzmann constant

T = gas temp

m_g = mass of gas molecule, H_2

assume p and T are power-laws in r :

$$p(r) = p_i \left(\frac{r_i}{r}\right)^k \quad T(r) = T_i \left(\frac{r_i}{r}\right)^{-l}$$

$r_i = 1 \text{ AU}$ p_i, T_i = values at $r = r_i$,

and $k, l > 0$

$$\text{so } p(r) \propto r^{-(k+l)}$$

$$\frac{dp}{dr} = -(k+l)p/r$$

so $T \sim 280K$ at $r=1\text{AU}$

The gas sound speed is

$$c = \sqrt{\frac{3k_B T}{m_g}} \quad m_g = 3.34 \times 10^{-24} \text{ gm}$$

$$k_B = 1.38 \times 10^{-16} \text{ erg/K}$$

$$\approx 3 \text{ km/sec}$$

while the circular Keplerian speed at $r=1\text{AU}$ is

$$v_{\text{KE}} = \sqrt{\frac{GM_p}{r}} \approx 30 \text{ km/sec}$$

a typical solar nebula model has $k \approx 3$, $f \sim 1$

(see reference at end)
of chapter 3

$$\text{so } \eta \approx 0.005$$

so the nebula gas' orbital velocity is sub-Keplerian by $\sim 0.5\%$, due to gas' radial pressure support which offsets (slightly)

orbiting
 \Rightarrow planetesimals feel headwind, which exerts gas-drag force on them

$$\text{and } \pi = \frac{(k+2)\rho}{2\rho(rB)^2}$$

for ideal gas,

ρ is also related to the nebula's sound speed c
via $\rho = \frac{1}{3}pc^2 = \rho_{\text{plast}}/f_{\text{pl}}$

$$\text{so } \pi = \frac{1}{6}(k+2) \left(\frac{c}{rB}\right)^2$$

To evaluate π , we need gas temp T .

The nebula is mostly H₂ gas (transparent)
+ ~1% dust by mass, so assume Sun
warms the dust which in turn warms H₂ gas
(naive assumption, but good enough for now)

The dust grains are blackbody radiators, so

$$\left(\frac{L_0}{4\pi r^2}\right) \pi r^2 = 4\pi r^2 \sigma T^4 = \text{rate at which blackbody radiates}$$

↓ ↓ ↓
 incident solar flux dust grain cross section area

$L_0 = 4 \times 10^{33} \text{ erg/sec}$
 $\sigma = 5.67 \times 10^{-5} \text{ erg/cm}^2 \text{ K}^4 \text{ sec}$

$$\text{so } T = \left(\frac{L_0}{16\pi r^2 \sigma}\right)^{1/4}$$

$$\sigma = 5.67 \times 10^{-5} \text{ erg/cm}^2 \text{ K}^4 \text{ sec}$$

asside: how thick (vertically) is the solar nebula at $r = 1 \text{ AU}$? ?

The gas molecules are in nearly Keplerian motion about Sun. The vertical velocity at a Keplerian orbit is

$$\dot{z} = a \sqrt{s} \sin I \cos(\omega t) \quad (\text{Eqn 2.51})$$

$$\text{so } c \sim a \sqrt{s} \sin I$$

↑
inclination of gas
molecule

$\frac{h}{r} = \text{nebula thickness}$
 $\text{nebula midplane} \rightarrow r$

$$\text{since } h = a \sin I$$

$$\text{then } c \approx h r \quad r = \sqrt{\frac{GM_0}{\mu^2}} = 2 \times 10^7$$

$$h = \frac{c}{r} \approx 0.1 \text{ AU}$$

so $\frac{h}{r} \sim 0.1 = \text{nebula's fractional thickness}$

drag acceleration on planetesimal due to

aerodynamic drag is

$$\vec{a}_d = - \frac{\rho}{2} C_d \frac{A}{m} \vec{u} u^2$$

\vec{u} = magnitude of \vec{u}

C_d = drag damping length

where $\vec{u} = \vec{r} - \vec{v}_{\text{gas}} = \text{planet's velocity}$
relative to gas

$$= \dot{r} \hat{r} + [r \dot{\theta} - (1-e) r \omega] \hat{\theta}$$

assume low-e motion: $\dot{r} = e a n \sin \delta$

$$\dot{\theta} = a n (1 - e \cos \delta)$$

note mean motion
 $n = 2\pi/T$ $r = a$

$$\text{so } \vec{u} = e a n \sin \delta \hat{r} + [a n + e a n \cos \delta - a n + n a n \delta] \hat{\theta}$$

$$\vec{u} = a n [e \sin \delta \hat{r} + (n + e \cos \delta) \hat{\theta}]$$

= planet's speed relative to gas

The relative speed is from

$$|u|^2 = (an)^2 \{ e^2 \sin^2 f + \pi^2 + 2en \cos f + e^2 \cos^2 f \}$$

$$= an \{ e^2 + \pi^2 + 2en \cos f \}$$

$$= an(e^2 + \pi^2) \left(1 + \frac{2en \cos f}{e^2 + \pi^2} \right)$$

$$\text{so } |u| = an \sqrt{e^2 + \pi^2} g(e/n, f)$$

$$\text{where } g(e/n, f) = \sqrt{1 + \frac{2en \cos f}{1 + (en)^2}}$$

≈ 1 for almost any value of e/n

(except that an take $0 \leq g \leq \sqrt{2}$ when $e \approx n$)

$$\text{assume } g \approx 1 \text{ so } \vec{a}_d \approx - \sqrt{e^2 + \pi^2} an \vec{u}$$

λ_d \downarrow
planetary volume density

$$\text{where } \lambda_d = \frac{8}{3\pi} \left(\frac{\rho_p}{\rho_g} \right) R_p$$

\downarrow drag coefficient \uparrow ρ_g density

λ_d is roughly the distance the planetesimal must travel to sweep out its own mass, in yrs

$$\text{So } \vec{\alpha}_d = -\frac{\vec{u}|\vec{u}|}{\lambda_d}$$

$$\approx -\frac{\sqrt{e + n^2}(an)^2}{\lambda} \left[e \sin f \hat{r} + (n + e \cos f) \hat{\theta} \right]$$

$$\text{set} = -\alpha_d an^2 \left[e \sin f \hat{r} + (n + e \cos f) \hat{\theta} \right]$$

↑
dimensionless drag constant

$$\alpha_d = \sqrt{e + n^2} \frac{a}{\lambda}$$

Note that when $n=0$, this drag force resembles problem #3 from midterm

what did that drag do?

what happens when $n > 0$?

Aerodynamic drag on planetesimal is

$$\vec{F}_d = -\frac{1}{2} \rho_{\text{gas}} C_D A \vec{v}_{\text{rel}}$$

$$\vec{v}_{\text{rel}} = \vec{v} - \vec{v}_{\text{gas}}$$

$$\text{where } \vec{v} = \dot{r} \hat{r} + r \dot{\theta} \hat{\theta} = (1-r) \vec{v}_{\text{rel}}(r)$$

\approx planet's velocity relative to \vec{v}_{gas}

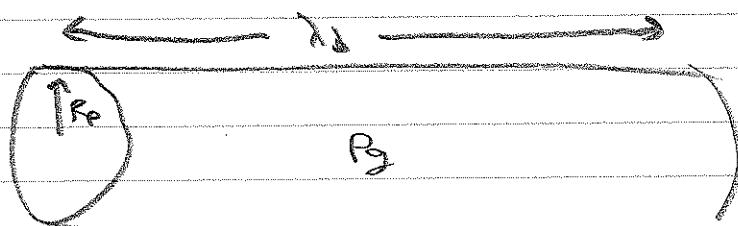
$$\text{and } |\vec{v}| = \rho_{\text{gas}} \sqrt{c_s^2 + v_r^2} \ g(e/m, f)$$

\approx constant

almost always 1

$$\lambda_d = \frac{2}{3} \left(\frac{\rho_p}{\rho_g} \right) R_p \sim \text{distance the planetesimal must travel to sweep a mass of gas equivalent to its own}$$

consider $R_p \approx 1 \text{ km}$ planetesimal



$$\rho_g \sim 10^{-9} \text{ gm/cm}^3 \quad (\text{see Section 3.2.23})$$

$$R_p \sim ?$$

$$\text{so } \lambda_d \sim 10^{15} \text{ cm} \sim 50 \text{ AU}$$

$$\text{if this planetesimal orbits at } r=1 \text{ AU, then } \lambda_d \sim \frac{50}{2\pi} \sim 10 \text{ orbits}$$

$$\text{set } \vec{a}_d = -\frac{|\vec{u}| \vec{u}}{\lambda_d} = -k_d \nu_0 \vec{u}$$

where $k_d = \frac{|\vec{u}|}{\lambda_d \nu_0} = \text{dimensionless drag coefficient}$

Note: $k_d \ll 1$ when drag is weak

$$\text{so } \vec{a}_d = -k_d \nu_0 \left[i \hat{i} + r (\dot{\theta} - \omega_0 + \nu_0 \omega) \hat{\theta} \right]$$

$$= -k_d \nu_0 \left[(i + r\dot{\theta}) \hat{i} + r (\omega_0 + \dot{\theta} + \dot{\omega}_0 - \nu_0 + \nu_0 \dot{\theta}) \hat{\theta} \right]$$

lets assume forcing by $M_g \gg$ that due to drag
(ie $|F_d| \ll |F_g|$ and $|\vec{a}_d| \ll |\vec{a}_g|$)

(check this assumption later)

$$\text{so } \vec{a}_d \approx -k_d \nu_0 i \hat{i} - k_d \nu_0 \omega_0 (\dot{\theta} + \nu_0 \omega_0) \hat{\theta}$$

is the linearized acceleration of particle due to gas drag
ie we also drop small nonlinear $r, \ddot{\theta}$, terms

we already wrote the EOM for a particle perturbed by m_s (see notes pg 81), so we just need to add \vec{a}_s to RHS:

$$\ddot{r} - r\dot{\theta}^2 = -\frac{d}{dr} \left[E_p + \gamma_m e^{im(\theta-\theta_0)} \right] - k_d r \omega_r,$$

and $\frac{dh}{dt} = -im\gamma_m e^{im(\theta-\theta_0)} - k_d r \omega_r (\dot{\theta}_0 + \gamma_m \omega_r)$

[the S EOM $\times \omega_r$ where $h = r^2\dot{\theta}$ = particle's specific angular momentum]

secular (steady) ang. mom. loss

but $\frac{dh}{dt} = h_i + h_o \leftarrow$ due to drag

\uparrow
due to m_s ,
is periodic

The oscillatory part and secular part of this EOM must be satisfied separately so

$$h_i = -im\gamma_m e^{i(m\theta_0 + \omega_r t)} - k_d r_0^2 \omega_r \dot{\theta},$$

and $h_o = -k_d \nabla (r_0 \omega_r)^2 =$ specific torque
(net gas drag exerts on particle.)

can integrate h_i :

$$h_i(t) = -\frac{m\gamma_m}{i\omega_r} e^{i(m\theta_0 + \omega_r t)} - k_d r_0^2 \omega_r \dot{\theta},$$

$$\text{Also } h = r^2 \dot{\theta} = (r_0 + r_1 + r_2)^2 (\omega_0 + \theta_1 + \theta_2)$$

$$= (r_0^2 + 2r_0r_1 + 2r_0r_2)(\omega_0 + \theta_1 + \theta_2)$$

$$= (r_0^2 \omega_0 + 2r_0 r_0 \omega_1 + r_0^2 \omega_2)$$

$$+ (2r_0 r_0 \theta_1 + r_0^2 \theta_2) \quad \text{when linearized}$$

(these are
semi-
non-oscillatory)

terms



periodic
term

$$\text{so } h = h_0(t) + h_1(t)$$

$$\text{where } h_0 = r_0^2 \omega_0 + 2r_0 r_0 \omega_1 + r_0^2 \omega_2$$

$$h_1(t) = 2r_0 r_0 \theta_1 + r_0^2 \theta_2$$

$$= -\frac{m}{w_n} \sin e^{i(m\theta_0 + w_n t)} - k_2 r_0^2 \omega_0 \theta_1$$

The oscillatory part of the particle's motion
will have the form

$$\eta(t) = -\operatorname{Re} [\operatorname{Re} e^{i(m\theta_0 + w_n t)}]$$

$$\theta_1(t) = \operatorname{Re} [\theta_1 e^{i(m\theta_0 + w_n t)}]$$

$$\text{so } \dot{\theta}_1 = i w_n \theta_1$$

$$2\omega_0 \sigma_r + i\omega_n \beta_0^2 \theta_i = -\frac{m}{\omega_n} \sin e^{i(\text{initial})} \\ - k_d r_0^2 \sigma_r \theta_i$$

$$(i\omega_n \beta_0^2 + k_d r_0^2 \sigma_r) \theta_i = - \left(\frac{m}{\omega_n} \sin e^{i\theta_i} + 2\sigma_r \theta_i \right)$$

$$\theta_i = \frac{1}{i\omega_n \beta_0^2 (1 + k_d r_0 / i\omega_n)} \left(\frac{m}{\omega_n} \sin e^{i\theta_i} + 2\sigma_r \theta_i \right)$$

$$= \frac{i}{\omega_n \beta_0^2} \left(1 - \frac{i k_d r_0}{\omega_n} \right)^{-1} \left(\frac{m}{\omega_n} \sin e^{i\theta_i} + 2\sigma_r \theta_i \right) \\ \uparrow \text{small since } k_d \ll 1$$

$$\text{so } \theta_i \approx \frac{i}{\omega_n \beta_0^2} \left(1 + \frac{i k_d r_0}{\omega_n} \right) \left[\frac{m}{\omega_n} \sin e^{i(\text{initial})} + 2\sigma_r \theta_i \right]$$

Next, solve for $r_i(t)$ using the \hat{r} part of the form